

Exam.Code:0940

Sub. Code: 7046

2010

**B.E. (Mechanical Engineering) Fourth Semester  
MEC-401: Applied Thermodynamics – II**

Time allowed: 3 Hours

Max. Marks: 50

*NOTE: Attempt five questions in all, including Question No. 1 which is compulsory and selecting two questions from each Unit.*

x-x-x

- I. Answers the following:-
- What is Prewhirl? Explain with the help of velocity triangles.
  - How Supercharging effect the efficiency of IC engines? What are superchargers?
  - How the IC engine fuels are rated?
  - Write the name of two solid propellants?
  - Which type of compression process uses minimum work? Sketch PV and TS diagram. (5x2)

**UNIT – I**

- II. a) Explain the stages of combustion in a CI engine.
- b) A four cylinder, four stroke engine is supplied with 15.25 kg of air/kg of octane fuel having heat value 10650 kcal/ kg. Calculate (i) The compression ratio. (ii) Brake specific fuel consumption (iii) bore and stroke; (iv) b.m.e.p. The following data may be assumed:- air standard efficiency: 0.53, relative efficiency: 0.70, mechanical efficiency: 0.8; volumetric efficiency: 0.8; stroke equal 1.25 times the bore; suction conditions are 1 kgf/cm<sup>2</sup> and 60 °C the engine runs at 2400 rev/min and develops 96 brake power.
- III. Air is drawn in a gas turbine unit at 15° C, 1.01 bar and pressure ratio is 7:1. The compressor is driven by the HP turbine and LP turbine drives a separate power shaft. The isentropic efficiencies of compressor, HP turbine and LP turbines are 0.82, 0.85 and 0.85 respectively. If the max cycle temperature is 610° C, calculate: pressure and temperature of the gases entering the power turbine, net power developed by the unit per kg/sec mass flow, work ratio and thermal efficiency of the unit. Neglect the mass of fuel and assume: for compression process  $C_{pa} = 1.005 \text{ KJ/KgK}$  and  $\gamma = 1.4$ , for combustion and expansion process:  $C_{pg} = 1.15 \text{ KJ/KgK}$  and  $\gamma = 1.333$ . (10)
- IV. The following data pertains to a turbojet flying an altitude of 9500 m: speed of turbojet = 800Km/hr, propulsive efficiency = 55%, overall efficiency of turbine plant = 17%, density of air at 9500 m altitude = 0.17 Kg/m<sup>3</sup>, drag on the plane 6100 N. Assuming calorific value of the fuels use as 46000 KJ/Kg calculate. Absolute velocity of the jet, volume of air compressed /Min, diameter of the jet, power output of the unit and air fuel ratio. (10)

P.T.O.

(2)

**UNIT – II**

- V. a) Explain the phenomenon of 'surging and choking' in a centrifugal compressor.  
b) Free air of  $30\text{m}^3/\text{min}$  is compressed from  $101.3\text{Kpa}$  to  $2.23\text{Bar}$ . Determine the power required if  
i) if roots blower is used.  
ii) if vane blower is used assume there is 25% reduction in volume before back flow occurs. Also find isentropic efficiency in each case. (2x5)
- VI. a) Explain the phenomena of slip and drive a relation for the calculation of slip factor using velocity triangles.  
b) Drive a relation for Euler's work in case of centrifugal compressor using velocity triangles. (2x5)
- VII. An axial flow compressor with an overall isentropic efficiency of 85% draws air at  $20^\circ\text{C}$  and compresses it in the pressure ratio of 4:1. The mean blade speed and flow velocity are constant throughout the compressor. Assuming 50% reaction blading and taking blade velocity as  $180\text{m}/\text{sec}$  and work input factor as 0.82. Calculate: Flow velocity and number of stages. Take  $\alpha_1 = 12^\circ$   $\beta_1 = 42^\circ$ . (10)

x-x-x