

(2)

- (b) The velocity potential function (ϕ) is given by:

$$\phi = -\frac{xy^3}{3} - x^2 + \frac{x^3y}{3} + y^2$$

Determine the velocity components in x and y directions and show that ϕ represents a possible case of flow. (6)

4. State Buckingham's π -theorem and obtain an expression for the force (**F**) exerted by a flowing fluid on a stationary body if it depends upon the length (**L**) of the body, velocity (**V**) of the fluid, density (**ρ**), viscosity (**μ**) of the fluid and acceleration due to gravity (**g**).

Section – C (Do any two questions)

5. (a) Differentiate between Minor and Major Losses. (4)
- (b) A pipe line carrying oil of specific gravity **0.8**, changes in diameter from **300 mm** at position **A** to **500 mm** at position **B** which is **5 m** at a higher level. If the pressures at **A** and **B** are **19.62 N/cm²** and **14.91 N/cm²** respectively, and the discharge is **150 litres/s**, determine the **loss of head** and **direction of flow**. (6)
6. What is Boundary layer. Why does it form over a solid surface.
For the velocity distribution in a boundary layer given by:

$$\frac{u}{U} = 2 \left(\frac{y}{\delta} \right) - \left(\frac{y}{\delta} \right)^2$$

where **u** is the velocity at a height **y** above the surface, **δ** is the nominal boundary layer thickness and **U** is the free stream velocity. Show that the ratio of boundary layer thickness to displacement thickness is equal to **3**. (3,7)

7. (a) A supersonic aircraft flies at an altitude of **1.8 km** where the temperature is **4°C**. The sound from the aircraft is heard by an observer on the ground **4** seconds after the aircraft passes directly overhead. Calculate the speed of the aircraft. Take **$\gamma = 1.4$** and **$R = 287 \text{ J/kg K}$** . (5)
- (b) Explain Reynolds experiment for determining the nature of flow in pipes. Define critical velocity and derive the expression for Reynolds number. (5)